

# FEES

FLEXIBLE EXPERIMENTAL EMBEDDED SATELLITE

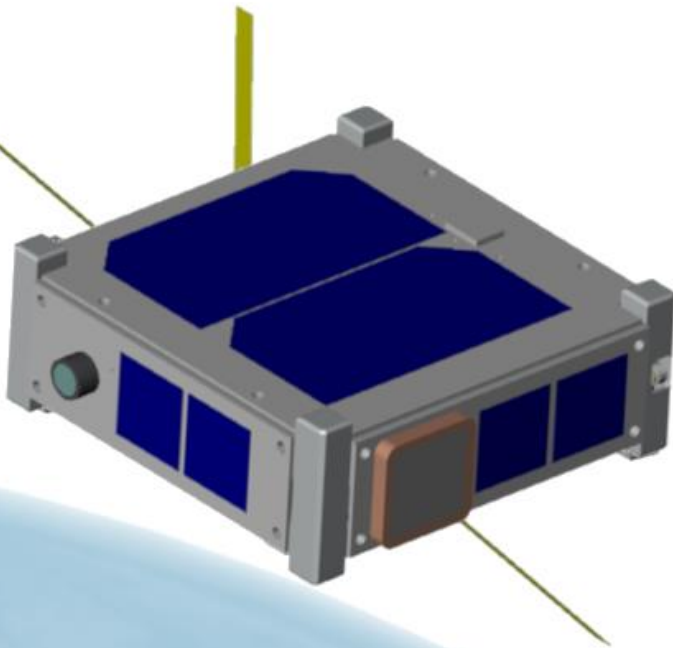
iCubeSat 28-29 May 2019

Politecnico di Milano

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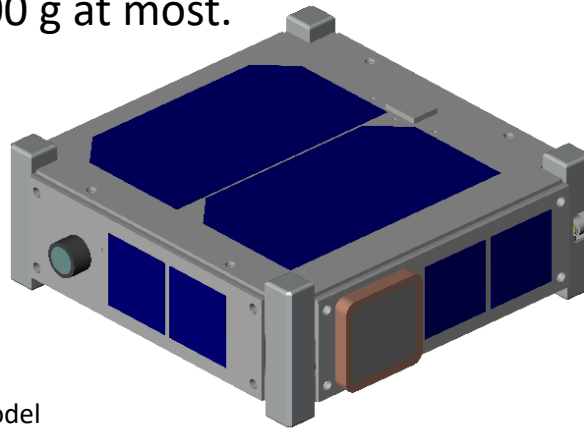
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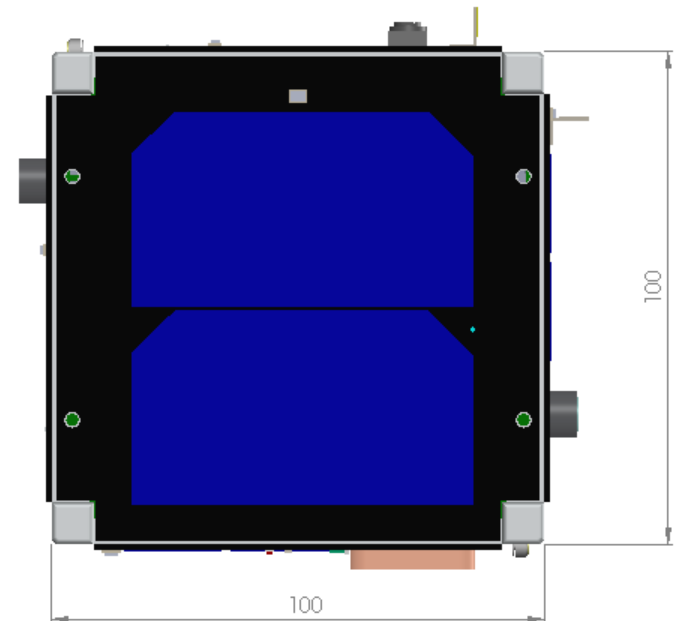
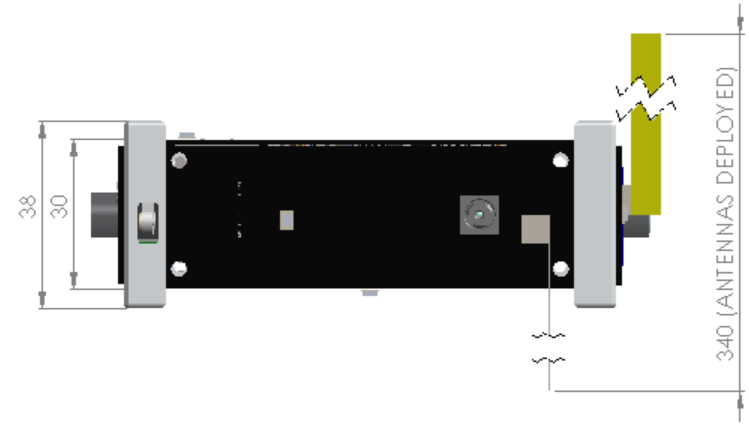


# Introduction

- **FEES** (Flexible Experimental Embedded Satellite) - a 1/3U CubeSat to be launched within Q1 2020 – is a **pico-platform** for technology IOV/IOT, focused on critical on-board subsystems, specifically miniaturized for such reduced platforms:
  - 10x10x3 cm volume
  - 300 g at most.



FEES model



# Partners

- GP Advanced Projects: project management and system engineering,
- Laser navigation: electronics and SDR, FW development,
- Politecnico di Milano: mission analysis, thermal analysis, ADCS algorithm, FW development, AIV/AIT,
- Linkit: TT&C,
- Brno University: total dose measurement,
- CESI: solar cells,
- Università degli Studi di Perugia: AIV/AIT.



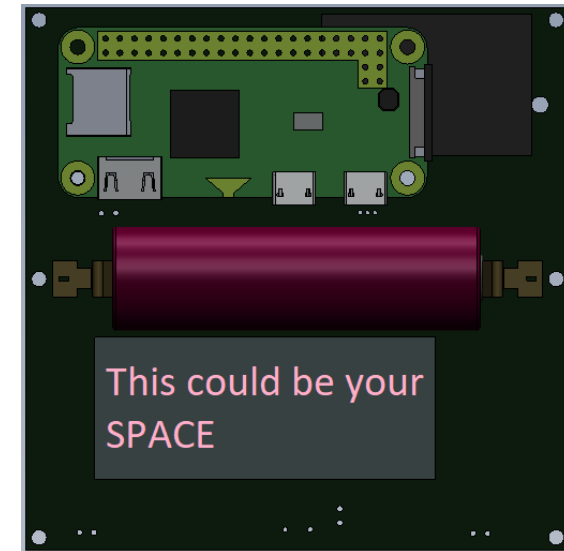
# Mission: statement and primary objective

« Flexible Experimental Embedded Satellite (FEES)'s goal is to validate an experimental platform for in-orbit testing of space components.»

- As previously introduced, the primary mission of the FEES is the **in-orbit testing and validation** of the S/C itself (e.g. Attitude Determination System) and the components defined as P/L. By doing so, FEES shall be considered a verified nano-platform for further launches, whether these will carry in-house designed components or stakeholder's P/L.
- Requirements
  - 2 weeks survival in SSO
  - Subsystem's testing
  - Data communication

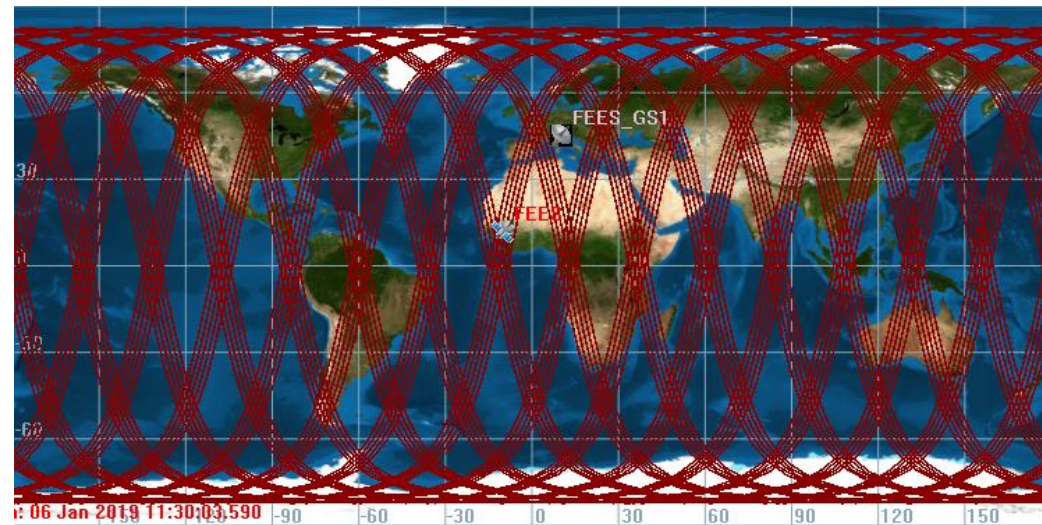
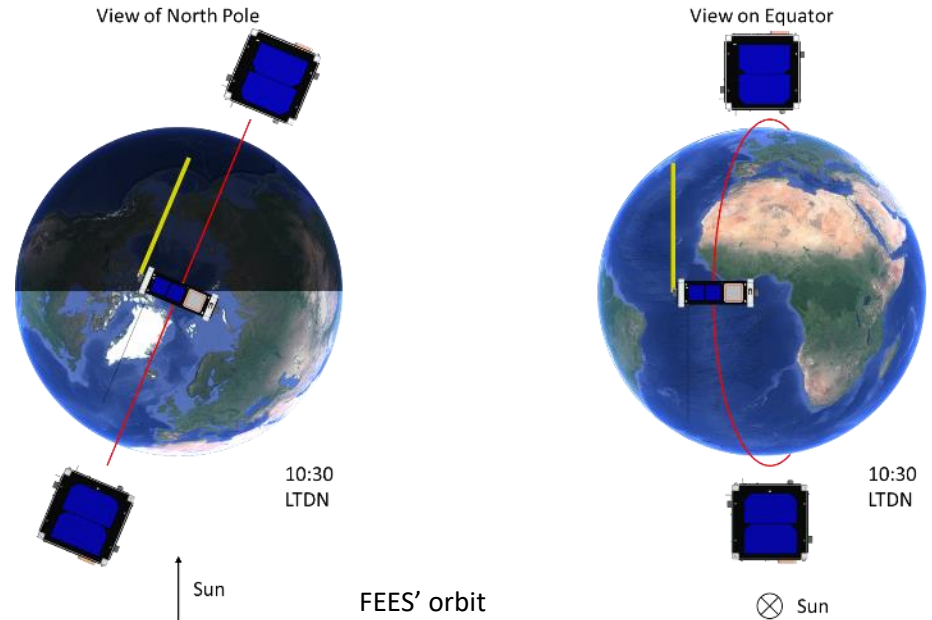
# Mission: secondary objectives

- Mission plan development:
  - extending the possibility of carrying third party P/L; availability onboard the next S/C for a low price.
- Data retrieving and handling:
  - Proving FEES is a useful, **versatile** platform for many LEO applications.
- Testing of Software Defined Radio (SDR) digital technology.
  - Different communication protocols will allow to identify the best solution for future missions' needs,



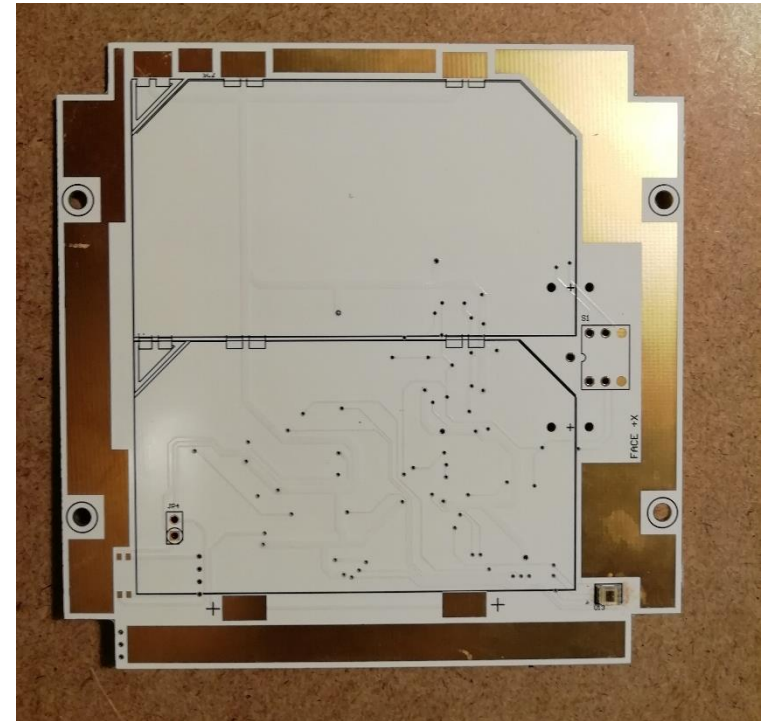
# Orbit

- Type: Sun-synchronous
- Altitude: ~575 km
- Inclination: 97°
- RAAN: 264°
- Period: 1,6 h
- Eccentricity: ~0
- Duration: 15 days



# Status

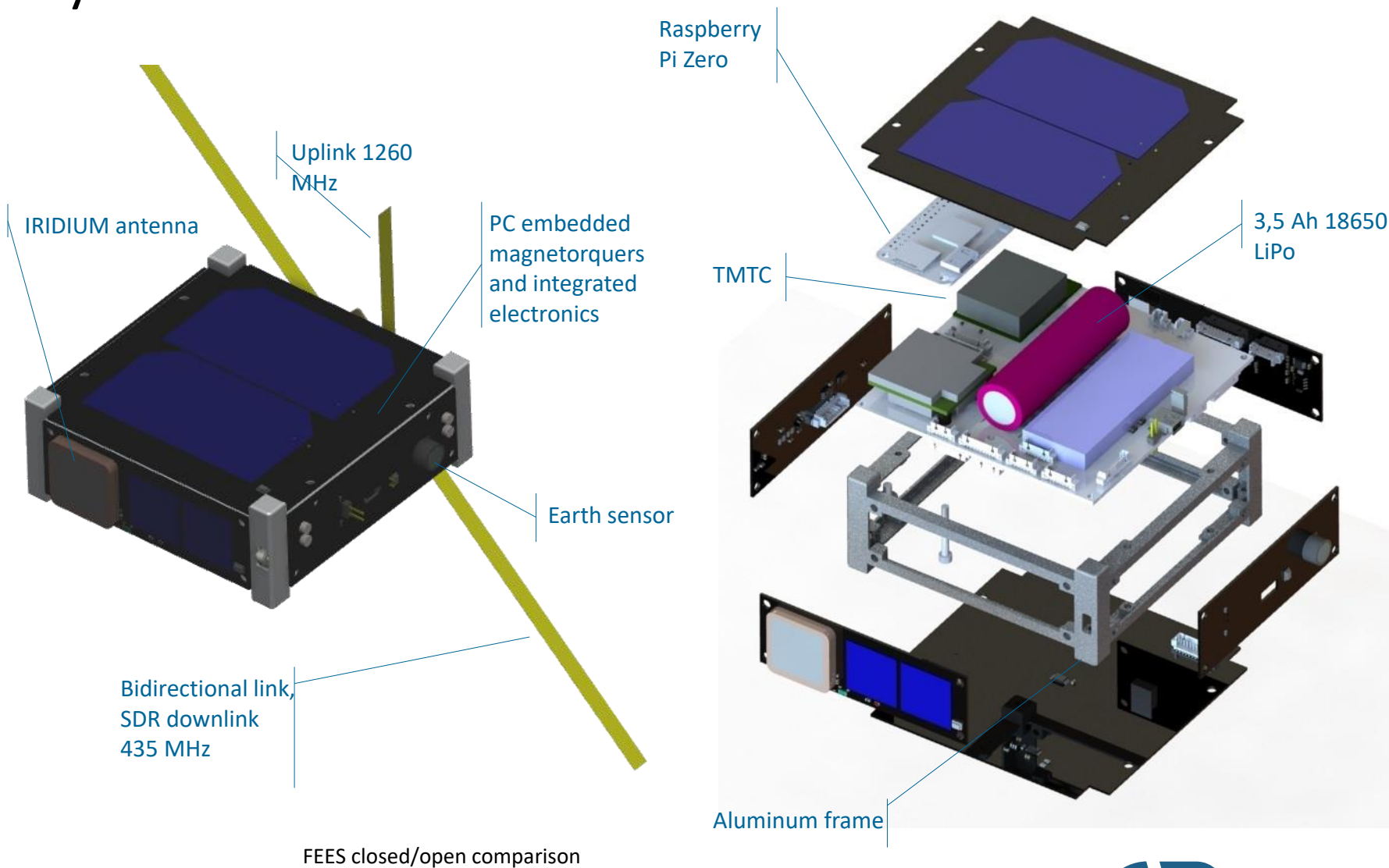
- PDR ✓
- QM – Manufacturing and Assemb ✓
  - Boards
  - Structure
  - EGSE/MGSE
- Test Campaign (Qualification - QM)
  - Late June 2019
- CDR
- Test Campaign (Acceptance - FM)
- Integration
- Launch (Soyuz-2, Q1 2020)



Top board



# Systems



29/05/2019

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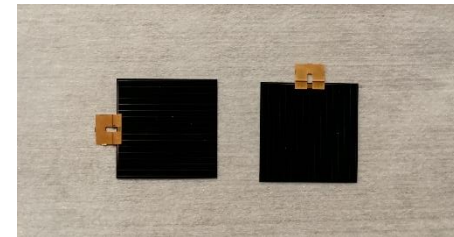


# Payload

- Complementary telecommunication system based on satellite calls constellation IRIDIUM (TBC regulations)
- RadEx2 (short for Radiation Experiment 2), a miniature-scale TID (total ionizing dose) experiment
- 2x2 cm experimental solar cells
- Software Defined Radio experiment
- Earth imaging camera



IRIDIUM module



2x2 cells



FEES' camera

# Thank you!

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